Chapter 1 Introduction

Processing ---- Structure -----

Semiconductors Advanced Bio Materials Material Smart Materials Nano Materials - e - wind wind with with the series سط رساختار تعيين لشرداس. (استذكى قلبى جنسير محانب الملك) درداداى خدان من سابق خرد برى دون - Piezoelectric (استذكى قلب) بالمعان در مناب المن مناب في المناب المن و برى دونا المند - Piezoelectric (سال المناب المن و برى دونا المند - Piezoelectric (سال المناب المماب المناب المناب المناب المناب المناب المناب المناب المنا Smart Material @ Magneto Strictive Material Electro rheological Huids العام سان معتملين واندر غلغ معتملين واندر عنف المعنام ال معام المعنام المع - Sensor + actuators ~ Critication - Sensor + + بن منه المنونيون + واغن (نانو تيوب + واغن



Chapter 2 Atomic Structure and Interatomic Bonding

 $E_N = \int_{-\infty}^{\infty} F_N \, dr$

 $F = \frac{dE}{dr}$

 $F_N = F_A + F_R$

 $=\frac{dE_A}{dr}+\frac{dE_R}{dr}$

 $= E_A + E_R$

 $= \int_{r}^{\infty} F_A \, dr + \int_{r}^{\infty} F_R \, dr$

The attractive bonding forces are coulombic

$$E_A = -\frac{A}{r}$$

Theoretically, the constant A is equal to

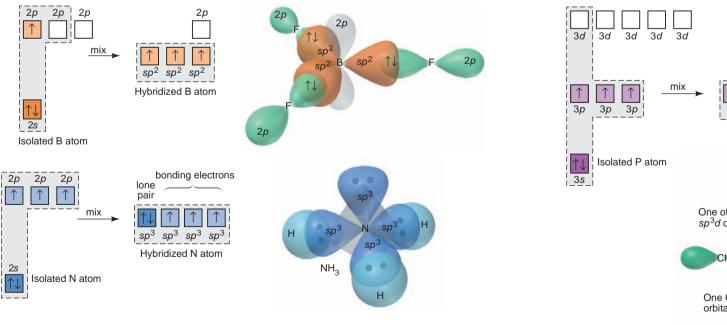
$$A = \frac{1}{4\pi\epsilon_0} (|Z_1|e) (|Z_2|e$$

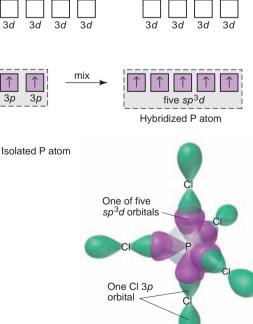
An analogous equation for the repulsive energy is⁵

$$E_R = \frac{B}{r^n}$$

Percent ionic character(%IC) of a bond between elements A and B :

 $\% IC = \left\{ 1 - e^{-\left(\frac{X_A - X_B}{2}\right)^2} \right\}$ $\times 100$





EXAMPLE PROBLEM 2.2

Computation of Attractive and Repulsive Forces between Two Ions

The atomic radii of K^+ and Br^- ions are 0.138 and 0.196 nm, respectively.

- (a) Using Equations 2.9 and 2.10, calculate the force of attraction between these two ions at their equilibrium interionic separation (i.e., when the ions just touch one another).
- (b) What is the force of repulsion at this same separation distance?

Solution

(a) From Equation 2.5b, the force of attraction between two ions is

$$F_A = \frac{dE}{dr}$$

Whereas, according to Equation 2.9,

$$E_A = -\frac{A}{r}$$

Now, taking the derivation of E_A with respect to r yields the following expression for the force of attraction F_A :

$$F_A = \frac{dE_A}{dr} = \frac{d\left(-\frac{A}{r}\right)}{dr} =$$

Now substitution into this equation the expression for A (Eq. 2.10) gives

$$F_A = \frac{1}{4\pi\epsilon_0 r^2} (|Z_1|e)$$

Incorporation into this equation values for *e* and ϵ_0 leads to

$$F_A = \frac{1}{4\pi (8.85 \times 10^{-12} \,\mathrm{F/m})(r^2)} \left[|Z_1| (1.602) + \frac{(2.31 \times 10^{-28} \,\mathrm{N \cdot m^2})(|Z_1|)(|Z_2|)}{r^2} \right]$$

For this problem, r is taken as the interionic separation r_0 for KBr, which is equal to the sum of the K⁺ and Br⁻ ionic radii inasmuch as the ions touch one another – that is,

$$r_0 = r_{K^+} + r_{Br^-}$$

= 0.138 nm + 0
= 0.334 nm

 $= 0.334 \times 10^{-9} \,\mathrm{m}$

When we substitute this value for r into Equation 2.14, and taking ion 1 to be K^+ and ion 2 as Br⁻ (i.e., $Z_1 = +1$ and $Z_2 = -1$), then the force of attraction is equal to

$$F_A = \frac{(2.31 \times 10^{-28} \,\mathrm{N \cdot m^2})(|+1|)(1.00 \,\mathrm{N \cdot m^2})}{(0.334 \times 10^{-9} \,\mathrm{m})^2}$$

(b) At the equilibrium separation distance the sum of attractive and repulsive forces is zero according to Equation 2.4. This means that

 $F_R = -F_A = -(2.07 \times 10^{-9} \text{ N}) = -2.07 \times 10^{-9} \text{ N}$

EXAMPLE PROBLEM 2.3

Calculation of the Percent Ionic Character for the C-H Bond

Compute the percent ionic character (%IC) of the interatomic bond that forms between carbon and hydrogen.

Solution

The %IC of a bond between two atoms/ions, A and B (A being the more electronegative) is a function of their electronegativities X_A and X_B , according to Equation 2.16. The electronegativities for C and H (see Figure 2.9) are $X_{\rm C} = 2.5$ and $X_{\rm H} = 2.1$. Therefore, the %IC is

$$%IC = \{1 - \exp[-(0.25)(X_{\rm C} - X_{\rm H})^2]\} \times 100$$
$$= \{1 - \exp[-(0.25)(2.5 - 2.1)^2]\} \times 100$$
$$= 3.9\%$$

Thus the C—H atomic bond is primarily covalent (96.1%).

$$-\left(\frac{-A}{r^2}\right) = \frac{A}{r^2} \tag{2.12}$$

$$(|Z_2|e) \tag{2.13}$$

 $\times 10^{-19}$ C)][$|Z_2|(1.602 \times 10^{-19}$ C)]

(2.14)

(2.15)

0.196 nm

 $\frac{(|-1|)}{2} = 2.07 \times 10^{-9} \,\mathrm{N}$

```
(X_{\rm C} - X_{\rm H})^2 × 100
```

Chapter 3 The Structure of **Crystalline Solids**



HCP

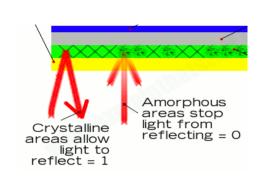
a=2R

 $V_C = \frac{3\sqrt{3}}{2}a^2C$

74%

6

12



Crystal Structures

Crystal System	Axial Relationships	Interaxial Angles	Unit Cell Geometry			
Cubic	a = b = c	$\alpha = \beta = \gamma = 90^{\circ}$			BCC $A = \frac{4R}{\sqrt{3}}$	FCC $A = \frac{4R}{\sqrt{2}}$
Hexagonal	$a = b \neq c$	$\alpha = \beta = 90^\circ, \gamma = 120^\circ$		Packing Factor N	68%	74%
Tetragonal	$a = b \neq c$	$\alpha = \beta = \gamma = 90^{\circ}$	c	Coordination	8	12
Rhombohedral (Trigonal)	a = b = c	$\alpha = \beta = \gamma \neq 90^{\circ}$			مرکز پر : enter وجوه پر : enter	
Orthorhombic	$a \neq b \neq c$	$\alpha = \beta = \gamma = 90^{\circ}$	c a b		ینی) پر : enter	
Monoclinic	$a \neq b \neq c$	$lpha=\gamma=90^\circ eqeta$		• ABCABC Stacking A sites B sites	acking Sequen Sequence	
Triclinic	$a \neq b \neq c$	$\alpha \neq \beta \neq \gamma \neq 90^{\circ}$	c a a	C sites • FCC Unit Cell		
			b 🕆 -			iso

	Simple	Body center	Face center	Base center
Cubic	\checkmark	\checkmark	\checkmark	×
Hexagonal	\checkmark	×	×	×
Tetragonal	\checkmark	×	×	×
Rhombohedral	\checkmark	×	×	×
Orthorhombic	\checkmark	\checkmark	\checkmark	\checkmark
Monoclinic	\checkmark	\checkmark	×	×
Triclinic	\checkmark	×	×	×

مرکز پایه ها (وج

Hexagonal Close-Packed Structure (HCP) • ABAB... Stacking Sequence



B sites

sites

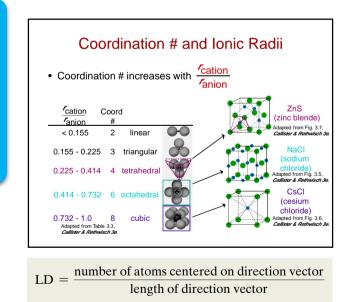


 V_{c}

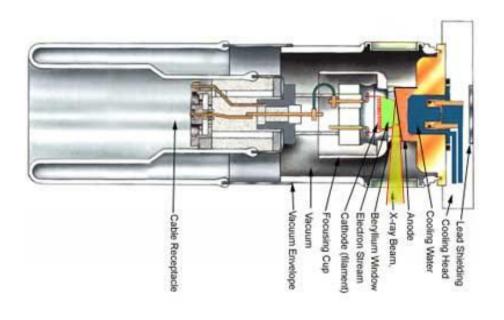
فواص در همه جوات یکسان : isotropic فواص در همه جوات غیریکسان: anisotropic

. اکر زرات بصورت کاملا رندوم قرار گرفته باشنر ،ماده ایزوتوپ است و اکرزرات بصورت فاص آرایش کرفته باشنر ماره anisotropic است . به عنوان مثال غور تک کریستال anisotropic ، BCC است .

B



number of atoms centered on a plane PD =area of plane

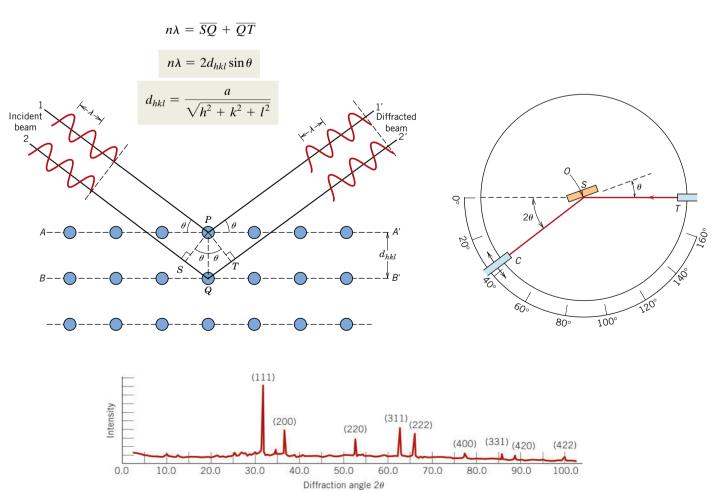


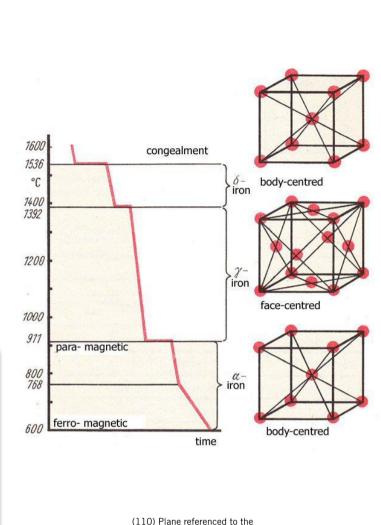


ما فقط یک طول میخواهیم پس از یک فیلتر استفاده میکنیم . عناصر موج هایی با طول موج های معینی را جذب میکنند و در جامدات آن ها بصورت پیوسته است ؛ یعنی **از طول موجی به قبل را در بازه ای جذب میکنند و از آن به بعد را جذب نمی کنند و این چنین ما فقط kaرا جدا می کنیم .**

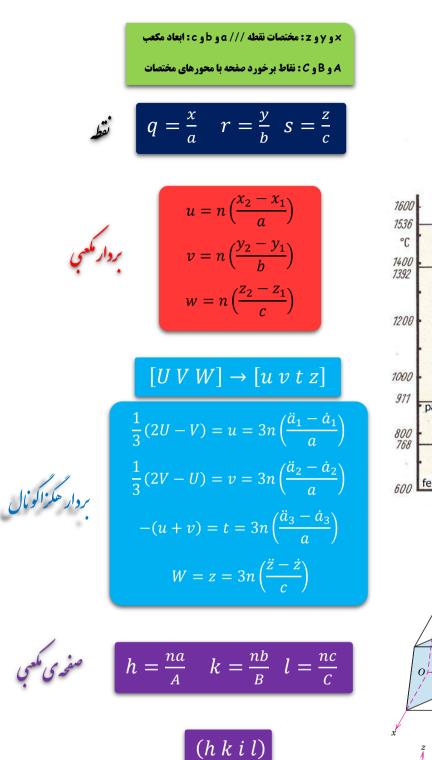
 $X - Ray \Rightarrow Fillter \Rightarrow تابش به بلور \Rightarrow Diffraction \Rightarrow شبت یک اثر انگشت برای ماره <math>\Rightarrow$ چرفش صفمه \Rightarrow

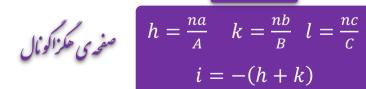


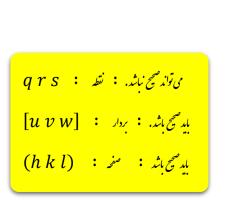


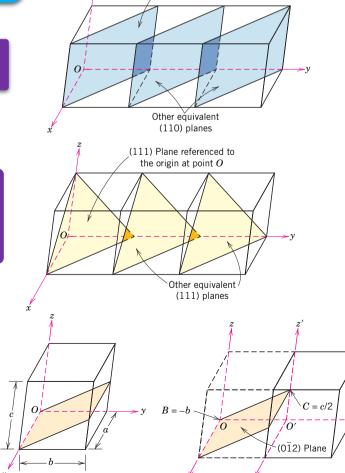


origin at point O



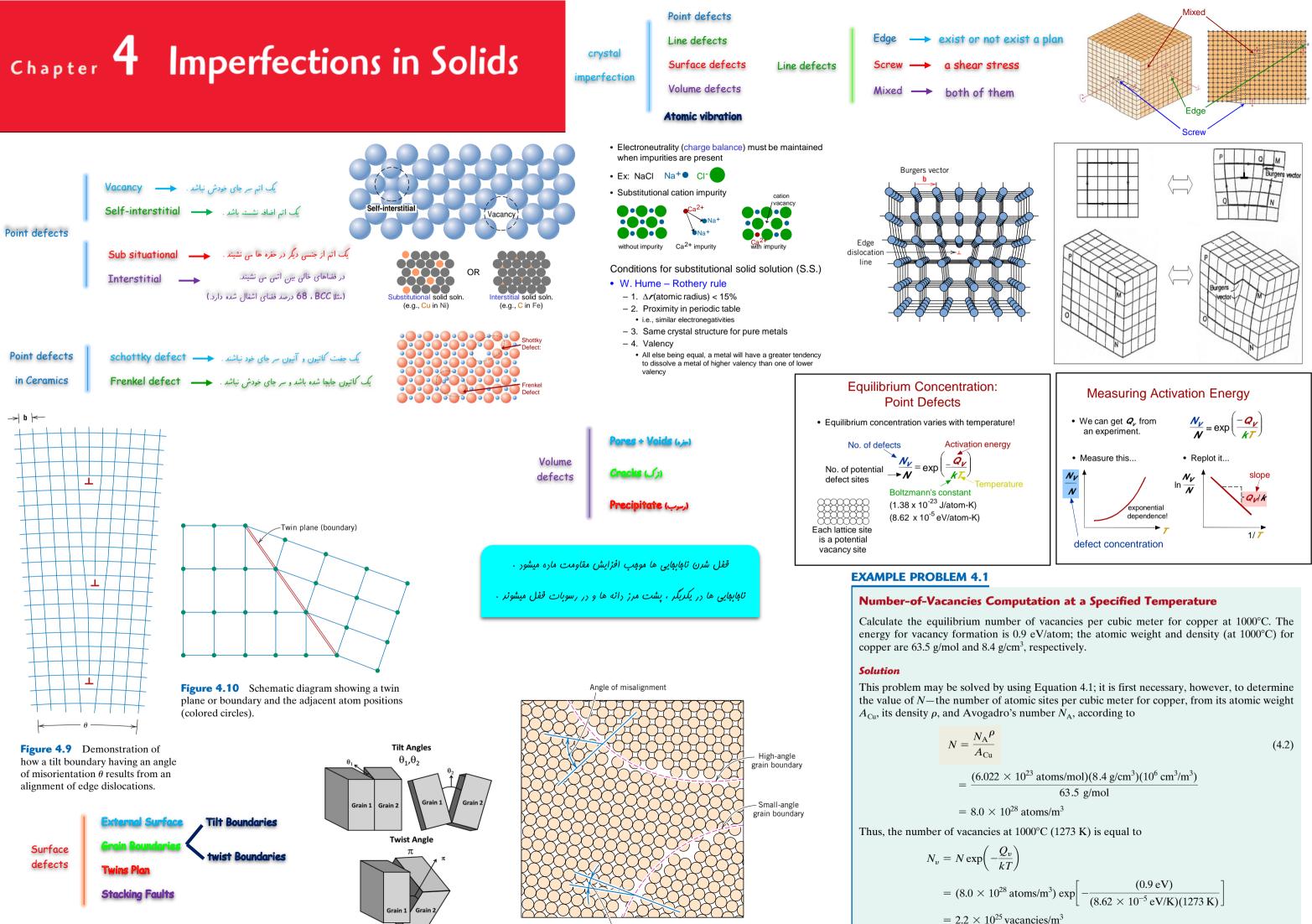






(b)

(a)



Angle of misalignment

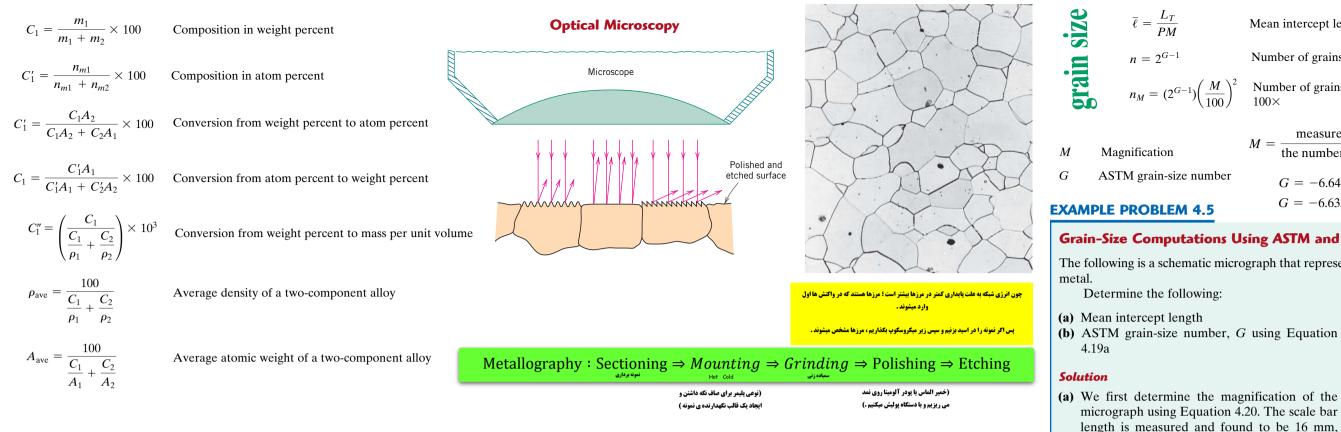
$$(4.2)$$

$$(22 \times 10^{23} \text{ atoms/mol})(8.4 \text{ g/cm}^3)(10^6 \text{ cm}^3/\text{m}^3))$$

$$(53.5 \text{ g/mol}) \times 10^{28} \text{ atoms/m}^3$$
es at 1000°C (1273 K) is equal to
$$-\frac{Q_v}{kT}$$

$$(0.9 \text{ eV})$$

$$(-\frac{1}{(8.62 \times 10^{-5} \,\mathrm{eV/K})(1273 \,\mathrm{K})}]$$



is

Line Number	Number of Grain- Boundary Intersections
1	8
2	8
3	8
4	9
5	9
6	9
7	7
Total	58

equal to

 $\bar{\ell} = \frac{L_T}{PM}$

therefore,

Mean intercept length (measure of average grain diameter)

Number of grains per square inch at a magnification of $100 \times$

$$\left(\frac{M}{100}\right)^2$$

Number of grains per square inch at a magnification other than
$$100 \times$$

$$M = \frac{\text{measured scale length (converted to microns)}}{\text{the number appearing by the scale bar (in microns)}}$$

$$G = -6.6457 \log \overline{\ell} - 3.298 \quad (\text{for } \overline{\ell} \text{ in mm})$$
$$G = -6.6353 \log \overline{\ell} - 12.6 \quad (\text{for } \overline{\ell} \text{ in in.})$$

Grain-Size Computations Using ASTM and Intercept Methods

The following is a schematic micrograph that represents the microstructure of some hypothetical

micrograph using Equation 4.20. The scale bar length is measured and found to be 16 mm, which is equal to 16,000 µm; and because the scale bar number is 100 µm, the magnification

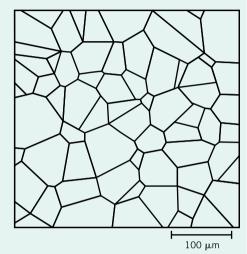
$$M = \frac{16,000 \,\mu\text{m}}{100 \,\mu\text{m}} = 160 \times$$

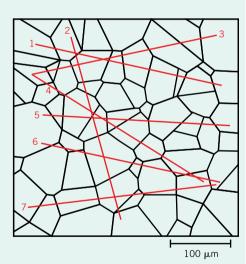
The following sketch is the same micrograph on which have been drawn seven straight lines (in red), which have been numbered.

The length of each line is 50 mm, and thus the total line length (L_T in Equation 4.16) is

(7 lines)(50 mm/line) = 350 mm

Tabulated next is the number of grain-boundary intersections for each line:





Thus, inasmuch as $L_T = 350$ mm, P = 58 grain-boundary intersections, and the magnification $M = 160 \times$, the mean intercept length $\overline{\ell}$ (in millimeters in real space), Equation 4.16, is

 $= \frac{350 \text{ mm}}{(58 \text{ grain boundary intersections})(160\times)}$ = 0.0377 mm

(b) The value of G is determined by substitution of this value for $\overline{\ell}$ into Equation 4.19a;

$$G = -6.6457 \log \ell - 3.298$$

= (-6.6457) log(0.0377) - 3.298
= 6.16

Chapter 6 Mechanical Properties of Metals

-orce F

 $\left(\frac{dF}{dr}\right)$

-400

400

300

200

100

0

-200

Strain

0.002

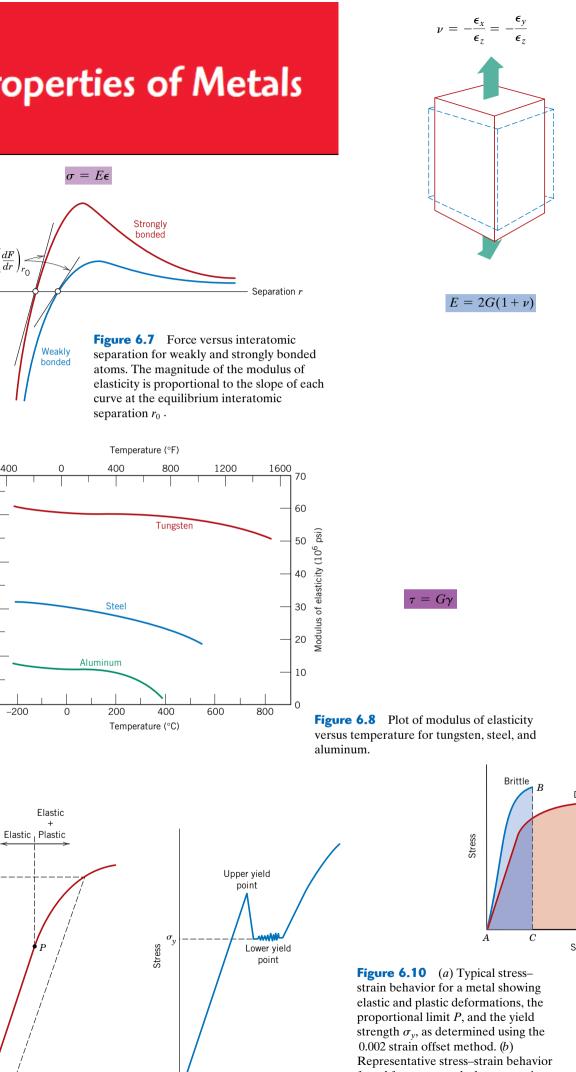
(GPa)

asti

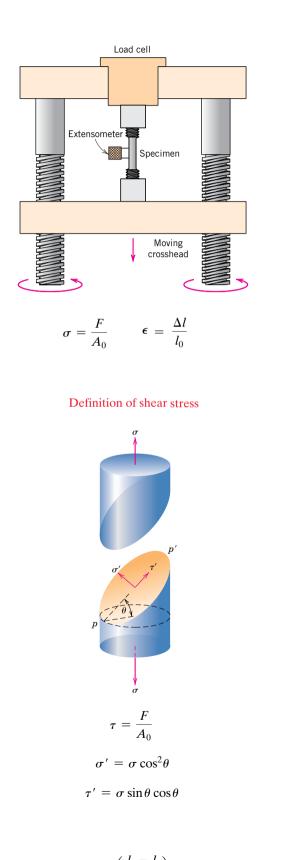
Ť

Modu

Stress



Strain



 $\% \text{EL} = \left(\frac{l_f - l_0}{l_0}\right) \times 100$ $\% \mathbf{RA} = \left(\frac{A_0 - A_f}{A_0}\right) \times 100$

EXAMPLE PROBLEM 6.2

A tensile stress is to be applied along the long axis of a cylindrical brass rod that has a diameter of 10 mm (0.4 in.). Determine the magnitude of the load required to produce a 2.5×10^{-3} mm $(10^{-4}$ -in.) change in diameter if the deformation is entirely elastic.

Solution

This deformation situation is represented in the accompanying drawing.

When the force *F* is applied, the specimen will elongate in the z direction and at the same time experience a reduction in diameter, Δd , of 2.5×10^{-3} mm in the x direction. For the strain in the x direction,

 ϵ_r

which is negative because the diameter is reduced. It next becomes necessary to calculate the strain in the z direction using Equation 6.8. The value for Poisson's ratio for brass is 0.34 (Table 6.1), and thus

 $\boldsymbol{\epsilon}_{z}$

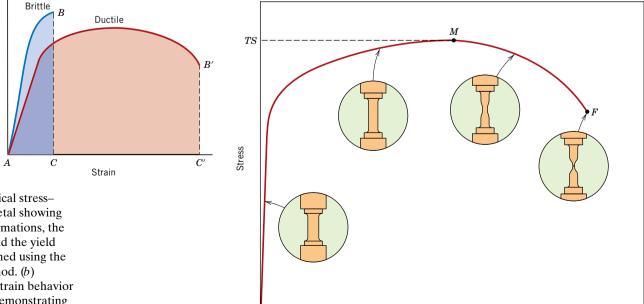
The applied stress may now be computed using Equation 6.5 and the modulus of elasticity, given in Table 6.1 as 97 GPa $(14 \times 10^6 \text{ psi})$, as

 $\sigma = \epsilon_{\tau}$

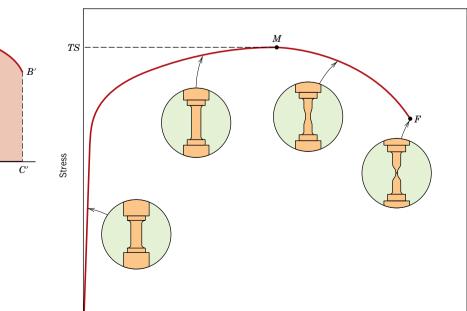
Finally, from Equation 6.1, the applied force may be determined as

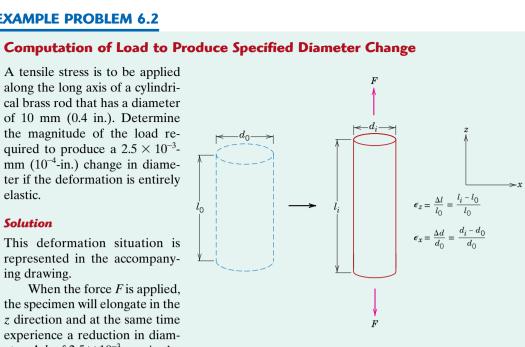
```
F = \sigma A_0
```

= (71.3



found for some steels demonstrating the yield point phenomenon.





$$=\frac{\Delta d}{d_0}=\frac{-2.5\times10^{-3}\,\mathrm{mm}}{10\,\mathrm{mm}}=-2.5\times10^{-4}$$

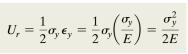
$$= -\frac{\epsilon_x}{\nu} = -\frac{(-2.5 \times 10^{-4})}{0.34} = 7.35 \times 10^{-4}$$

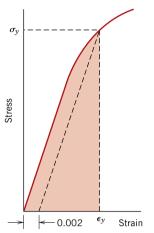
$$E = (7.35 \times 10^{-4})(97 \times 10^{3} \text{ MPa}) = 71.3 \text{ MPa}$$

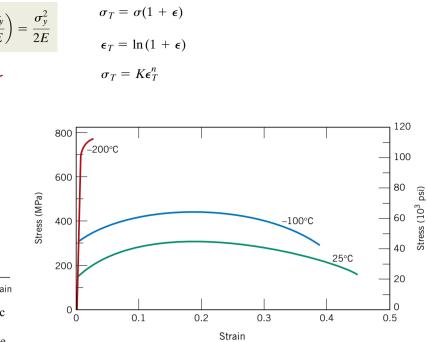
$$= \sigma \left(\frac{d_0}{2}\right)^2 \pi$$

$$\times 10^6 \,\text{N/m}^2 \left(\frac{10 \times 10^{-3} \,\text{m}}{2}\right)^2 \pi = 5600 \,\text{N} (1293 \,\text{lb}_f)$$

Resilience



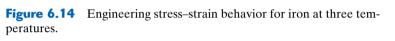




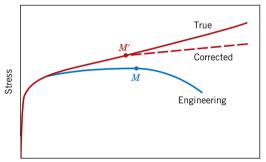
osi)

 $\boldsymbol{\epsilon}_T = \ln \frac{l_i}{l_0}$

Figure 6.15 Schematic representation showing how modulus of resilience (corresponding to the shaded area) is determined from the tensile stress-strain behavior of a material.

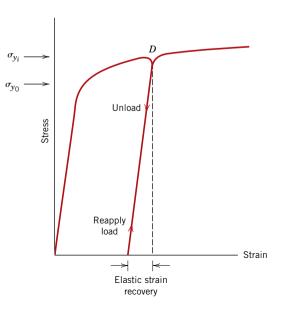


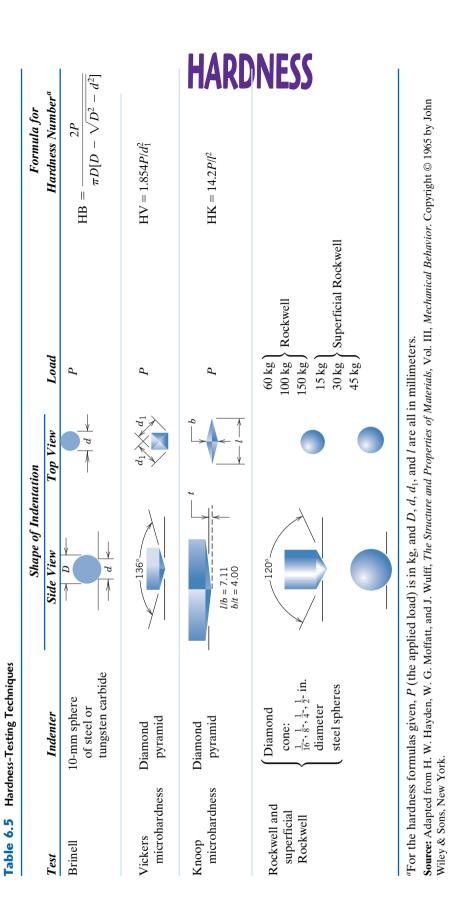
TRUE STRESS AND STRAIN



Strain

Figure 6.16 A comparison of typical tensile engineering stress-strain and true stress-strain behaviors. Necking begins at point *M* on the engineering curve, which corresponds to M' on the true curve. The "corrected" true stress-strain curve takes into account the complex stress state within the neck region.





EXAMPLE PROBLEM 6.4

Ductility and True-Stress-at-Fracture Computations

Ductility and True-Stre
A cylindrical specimen of a tested to fracture and found If its cross-sectional diamet
(a) The ductility in terms of(b) The true stress at fract
Solution
(a) Ductility is computed u
% R.
(b) True stress is defined by area A_f . However, the l $F = \sigma_f A_0 =$
Thus, the true stress is a
σ_T =
=
EXAMPLE PROBLEM 6.
Calculation of Strain-H
Compute the strain-harden of 415 MPa (60,000 psi) profor <i>K</i> .
Solution This requires some algebra parameter. This is accompli

п

steel having an original diameter of 12.8 mm (0.505 in.) is tensiled to have an engineering fracture strength σ_f of 460 MPa (67,000 psi). ter at fracture is 10.7 mm (0.422 in.), determine

of percentage reduction in area ure

using Equation 6.12, as

$$A = \frac{\left(\frac{12.8 \text{ mm}}{2}\right)^2 \pi - \left(\frac{10.7 \text{ mm}}{2}\right)^2 \pi}{\left(\frac{12.8 \text{ mm}}{2}\right)^2 \pi} \times 100$$
$$= \frac{128.7 \text{ mm}^2 - 89.9 \text{ mm}^2}{128.7 \text{ mm}^2} \times 100 = 30\%$$

y Equation 6.15, where, in this case, the area is taken as the fracture oad at fracture must first be computed from the fracture strength as

$$(460 \times 10^6 \,\mathrm{N/m^2})(128.7 \,\mathrm{mm^2}) \left(\frac{1 \,\mathrm{m^2}}{10^6 \,\mathrm{mm^2}}\right) = 59,200 \,\mathrm{N}$$

calculated as

$$= \frac{F}{A_f} = \frac{59,200 \text{ N}}{(89.9 \text{ mm}^2) \left(\frac{1 \text{ m}^2}{10^6 \text{ mm}^2}\right)}$$
$$= 6.6 \times 10^8 \text{ N/m}^2 = 660 \text{ MPa} (95,700 \text{ psi})$$

5

Hardening Exponent

ning exponent *n* in Equation 6.19 for an alloy in which a true stress oduces a true strain of 0.10; assume a value of 1035 MPa (150,000 psi)

aic manipulation of Equation 6.19 so that n becomes the dependent lished by taking logarithms and rearranging. Solving for *n* yields

$$= \frac{\log \sigma_T - \log K}{\log \epsilon_T}$$
$$= \frac{\log(415 \text{ MPa}) - \log(1035 \text{ MPa})}{\log(0.1)} = 0.40$$

EXAMPLE PROBLEM 6.6

Average and Standard Deviation Computations

The following tensile strengths were measured for four specimens of the same steel alloy:

Sample Number	Tensile Strength (MPa)
1	520
2	512
3	515
4	522

(a) Compute the average tensile strength.

(b) Determine the standard deviation.

Solution

(a) The average tensile strength (\overline{TS}) is computed using Equation 6.21 with n = 4:

$$\overline{TS} = \frac{\sum_{i=1}^{7} (TS)_i}{4}$$
$$= \frac{520 + 512 + 515 + 522}{4}$$

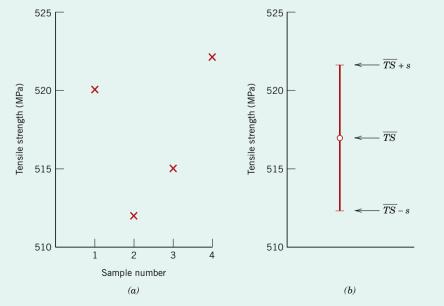
= 517 MPa

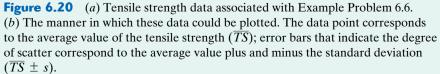
(b) For the standard deviation, using Equation 6.22, we obtain

$$s = \left[\frac{\sum_{i=1}^{4} \{(TS)_i - \overline{TS}\}^2}{4 - 1}\right]^{1/2}$$
$$= \left[\frac{(520 - 517)^2 + (512 - 517)^2 + (515 - 517)^2 + (522 - 517)^2}{4 - 1}\right]^{1/2}$$
$$= 4.6 \text{ MPa}$$

Figure 6.20 presents the tensile strength by specimen number for this example problem and also how the data may be represented in graphical form. The tensile strength data point (Figure 6.20*b*) corresponds to the average value \overline{TS} , and scatter is depicted by error

bars (short horizontal lines) situated above and below the data point symbol and connected to this symbol by vertical lines. The upper error bar is positioned at a value of the average value plus the standard deviation $(\overline{TS} + s)$, and the lower error bar corresponds to the average minus the standard deviation $(\overline{TS} - s)$.





Correlation between Hardness and Tensile Strength

$$TS(MPa) = 3.45 \times HB$$

 $TS(psi) = 500 \times HB$

Computation of Average and Standard Deviation Values

$$\overline{x} = \frac{\sum_{i=1}^{n} x_i}{n} \qquad s = \left[\frac{\sum_{i=1}^{n} (x_i - \overline{x})^2}{n-1}\right]^{1/2}$$

DESIGN/SAFETY FACTORS

$$\begin{array}{ll} \text{design stress} & \text{safe stress} \\ \sigma_d = N' \sigma_c & \sigma_w = \frac{\sigma_y}{N} \end{array}$$

DESIGN EXAMPLE 6.1

Specification of Support-Post Diameter

A tensile-testing apparatus is to be constructed that must withstand a maximum load of 220,000 N (50,000 lb_f). The design calls for two cylindrical support posts, each of which is to support half of the maximum load. Furthermore, plain-carbon (1045) steel ground and polished shafting rounds are to be used; the minimum yield and tensile strengths of this alloy are 310 MPa (45,000 psi) and 565 MPa (82,000 psi), respectively. Specify a suitable diameter for these support posts.

Solution

The first step in this design process is to decide on a factor of safety, N, which then allows determination of a working stress according to Equation 6.24. In addition, to ensure that the apparatus will be safe to operate, we also want to minimize any elastic deflection of the rods during testing; therefore, a relatively conservative factor of safety is to be used, say N = 5. Thus, the working stress σ_w is just

$$\sigma_w = \frac{\sigma_y}{N}$$
$$= \frac{310 \text{ MPa}}{5} = 62 \text{ MPa (9000 ps)}$$

From the definition of stress, Equation 6.1,

$$A_0 = \left(\frac{d}{2}\right)^2 \pi = \frac{F}{\sigma_w}$$

where d is the rod diameter and F is the applied force; furthermore, each of the two rods must support half of the total force, or 110,000 N (25,000 psi). Solving for d leads to

$$d = 2\sqrt{\frac{F}{\pi\sigma_w}}$$

= $\sqrt{\frac{110,000 \text{ N}}{\pi(62 \times 10^6 \text{ N/m}^2)}}$
= 4.75 × 10⁻² m = 47.5 mm (1.87 in

Therefore, the diameter of each of the two rods should be 47.5 mm, or 1.87 in.

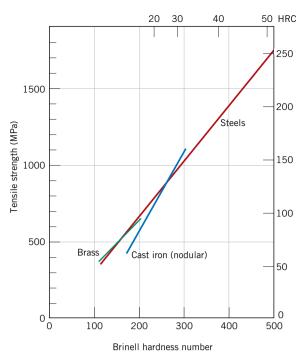
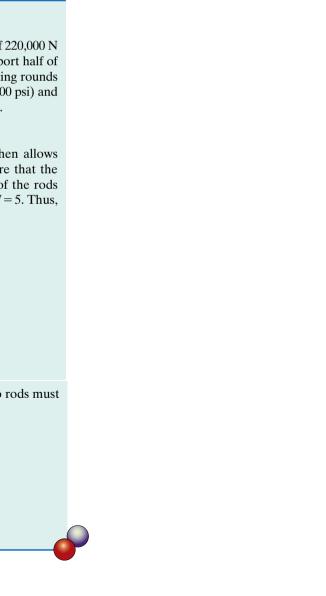


Figure 6.19 Relationships between hardness and tensile strength for steel, brass, and cast iron.



DESIGN EXAMPLE 6.2

tube with the lowest cost.

Materials Specification for a Pressurized Cylindrical Tube

(a) Consider a thin-walled cylindrical tube having a radius of 50 mm and wall thickness 2 mm that is to be used to transport pressurized gas. If inside and outside tube pressures are 20 and 0.5 atm (2.027 and 0.057 MPa), respectively, which of the metals and alloys listed in Table 6.8 are suitable candidates? Assume a factor of safety of 4.0.

For a thin-walled cylinder, the circumferential (or "hoop") stress (σ) depends on pressure difference (Δp), cylinder radius (r_i), and tube wall thickness (t) as follows:

$$r = \frac{r_i \,\Delta p}{t} \tag{6.25}$$

These parameters are noted on the schematic sketch of a cylinder presented in Figure 6.21. (b) Determine which of the alloys that satisfy the criterion of part (\mathbf{a}) can be used to produce a

Solution

(a) In order for this tube to transport the gas in a satisfactory and safe manner, we want to minimize the likelihood of plastic deformation. To accomplish this, we replace the circumferential stress in Equation 6.25 with the yield strength of the tube material divided by the factor of safety, *N*—that is,

$$\frac{\sigma_y}{N} = \frac{r_i \,\Delta p}{t}$$

And solving this expression for σ_v leads to

$$\sigma_y = \frac{Nr_i\,\Delta p}{t} \tag{6.26}$$

Table 6.8Yield Strengths, Densities, and Costs per Unit Mass for MetalAlloys That Are the Subjects of Design Example 6.2

Alloy	Yield Strength, σ_y (MPa)	Density, ρ (g/cm ³)	Unit mass cost, c (\$US/kg)
Steel	325	7.8	1.75
Aluminum	125	2.7	5.00
Copper	225	8.9	7.50
Brass	275	8.5	10.00
Magnesium	175	1.8	12.00
Titanium	700	4.5	85.00

We now incorporate into this equation values of $N, r_i, \Delta p$, and t given in the problem statement and solve for σ_y . Alloys in Table 6.8 that have yield strengths greater than this value are suitable candidates for the tubing. Therefore,

$$\sigma_y = \frac{(4.0)(50 \times 10^{-3} \text{ m})(2.027 \text{ MPa} - 0.057 \text{ MPa})}{(2 \times 10^{-3} \text{ m})} = 197 \text{ MPa}$$

(6.27)

Four of the six alloys in Table 6.8 have yield strengths greater than 197 MPa and satisfy the design criterion for this tube — that is, steel, copper, brass, and titanium.

(b) To determine the tube cost for each alloy, it is first necessary to compute the tube volume V, which is equal to the product of cross-sectional area A and length L-that is,

$$V = AL$$

= $\pi (r_o^2 - r_i^2)L$

Here, r_o and r_i are, respectively, the tube inside and inside radii. From Figure 6.21, it may be observed that $r_o = r_i + t$, or that

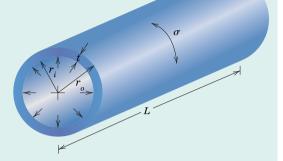


Figure 6.21 Schematic representation of a cylindrical tube, the subject of Design Example 6.2.

$$= \pi (r_o^2 - r_i^2)L = \pi [(r_i + t)^2 - r_i^2]L$$

= $\pi (r_i^2 + 2r_i t + t^2 - r_i^2)L$
= $\pi (2r_i t + t^2)L$ (6.28)

Because the tube length L has not been specified, for the sake of convenience, we assume a value of 1.0 m. Incorporating values for r_i and t, provided in the problem statement leads to the following value for V:

$$V = \pi [(2)(50 \times 10^{-3} \text{ m})(2 \times 10^{-3} \text{ m}) + (2 \times 10^{-3} \text{ m})^2](1 \text{ m})$$

= 6.28 × 10⁻⁴ m³ = 628 cm³

Next, it is necessary to determine the mass of each alloy (in kilograms) by multiplying this value of V by the alloy's density, ρ (Table 6.8) and then dividing by 1000, which is a unit-conversion factor because 1000 mm = 1 m. Finally, cost of each alloy (in \$US) is computed from the product of this mass and the unit mass cost (\bar{c}) (Table 6.8). This procedure is expressed in equation form as follows:

$$Cost = \left(\frac{V\rho}{1000}\right)(\bar{c}) \tag{6.29}$$

For example, for steel,

Cost (steel) =
$$\left[\frac{(628 \text{ cm}^3)(7.8 \text{ g/cm}^3)}{(1000 \text{ g/kg})}\right](1.75 \text{ $US/kg)} = \text{$8.60}$$

Cost values for steel and the other three alloys, as determined in the same manner are tabulated below.

Alloy	Cost (\$US)	
Steel	8.60	
Copper	41.90	
Brass	53.40	
Titanium	240.20	

Hence, steel is by far the least expensive alloy to use for the pressurized tube.